

ON THE USE OF ROBOTICS FORMALISM IN  
THE DESCRIPTION AND MODELING OF  
ANTENNA RANGE POSITIONERS

R. J. Pogorzelski and R. J. Beckon  
Mail Stop 138-307  
Jet Propulsion Laboratory  
California Institute of Technology  
4800 Oak Grove Drive  
Pasadena, CA 91109-8099

A typical positioner used for positioning an antenna under test on an antenna range has two or three rotation axes arranged in such a manner **as to facilitate the taking of data along certain paths through the antenna pattern (pattern cuts)**. It will usually have one horizontal rotation axis (the elevation axis), **a vertical axis (the azimuth axis)**, and possibly an additional axis (**sometimes called the roll axis**) which is oriented by the other two. **In the most straightforward of measurement sequences**, all but one of the axes are **fixed** and the desired axis is rotated over a prescribed range of angles during which data samples of the received signal are taken. The transmitting antenna **illuminates the antenna** under test **from a location in the far zone** resulting in a plane wave at the positioner. As measurement sequences **become more complex**, perhaps requiring simultaneous motion of several axes, two needs arise. **First**, given the desired angular trajectory, one must determine the necessary axis rotations to achieve it. **Second**, if phase data is to be properly **interpreted**, one must **obtain**, for the particular trajectory **used**, the transmitting antenna / receiving antenna distance for each **data** sample. These needs can be conveniently met by means of the existing formalism developed for description and control of the behavior of industrial robots. [See, for example, J. J. Craig, Introduction to Robotics, Addison-Wesley, Reading, MA, 1986]

To properly describe the behavior of a given positioner, one must first **identify** the parameters of the positioner model. We have done this by performing a set of diagnostic phase measurements which effectively measure the transmitting antenna / receiving antenna distance and optimizing the fit between these data and corresponding data generated by a simulated positioner. The fit is optimized by adjusting the geometrical parameters in the positioner model such as the angles between **the** various rotation axes. Once this parameter identification has been successfully accomplished, the positioner model can be used to prescribe the necessary axis rotations to achieve a desired trajectory and to determine the transmitter / receiver **distance** for each data point. In fact, even if the positioner is damaged or improperly constructed so as to render, for example, the angle between two ostensibly orthogonal axes significantly different from ninety degrees, robotics modeling can be used to determine the rotations necessary to compensate for the positioner shortcomings and achieve the desired trajectory. **In this paper** we describe our experiences with this application of robotics.

**JET PROPULSION LABORATORY**  
**NOTIFICATION OF CLEARANCE**

09/16/97

**TO: R. Pogorzelski**  
**FROM: Logistics and Technical Information Division**  
**SUBJECT: Notification of Clearance - CL 97-1197**

**The following title has been cleared by the Document Review Services, Section 644, for public release, presentation, and/or printing in the open literature:**

**On the Use of Robotics Formalism in the Description and Modeling of Antenna, Range Positioners**

**This clearance is issued for an abstract and is valid only for release in the U.S.**

**This clearance is issued for the abstract only; the full paper must be reviewed and approved prior to presentation or publication.**

Clearance issued by Charlotte Marsh  
Charlotte Marsh  
Document Review Services  
Section 644

(Over)



# National Radio Science Meeting

January 5-9, 1998  
The University of Colorado at Boulder



October 31, 1997

Mr. Ronald Pogorzelski  
Mail Stop 138-307  
Jet Propulsion Laboratory  
California Institute of Technology  
4800 Oak Grove Drive  
Pasadena, CA 91109-8099  
USA

Dear Colleague:

I am happy to inform you that your paper entitled:

*On the Use of Robotics Formalism in the Description and Modeling of Antenna Range Positioners*

has been accepted for presentation at the 1998 National Radio Science Meeting. It has been scheduled for Session A3-2 on 1/6/98 at 14:00.

All papers (except those 'scheduled for Friday, January 9) are scheduled in twenty-minute (or in rare instances thirty- or forty-minute) blocks. You should plan to **complete** your comments well within that time to **allow** the audience the opportunity to **ask questions** immediately following your presentation. .

A copy of the Advance Program for the meeting will be sent in November. It will include all program information as well as registration forms and travel and lodging information. The program will be available on the web at [http://cires. Colorado. edu/urstimprogram](http://cires.colorado.edu/urstimprogram) after November 10. If you have questions concerning your presentation, the local organizers, Rod Frehlich or Denise Thorsen, can be reached by e-mail at [ursi@cires.colorado.edu](mailto:ursi@cires.colorado.edu).

Thank you for your support of URSI and the 1998 National Radio Science Meeting.

Sincerely,

*Technical Program Committee*



## **On the Use of Robotics Formalism in the Description and Modeling of Antenna Range Positioners**

R. J. Pogorzelski and R. J. Beckon  
Jet Propulsion Laboratory  
California Institute of Technology  
4800 Oak Grove Drive  
Pasadena, CA 91109-8099

The formalism developed over the past few decades ~~for the~~ analytical and **numerical treatment of industrial robots provides a convenient tool for** calibration and **programming** of antenna range positioners. The existence of this formalism enables the application of standard system identification techniques such as the Levenburg-Marquardt algorithm to **derive** positioner **geometry** from range phase measurements. Once the positioner geometry is known, robotics can be used to program any desired trajectory and to implement compensation for geometrical defects in the positioner as well as compensation for range variation in phase measurements.



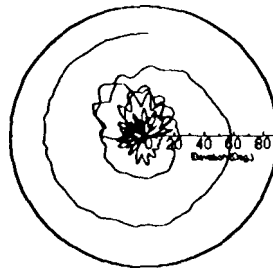
## Overview

- Motivated by the “Virtual Spacecraft Concept.”
  - Provides for “Antenna in the Loop” link simulation.
  - Antenna on a test range controlled by FST.
  - Requires complex positioner motion.
- Positioner geometry needed to enable the simulation.
  - Diagnostics
  - Compensation

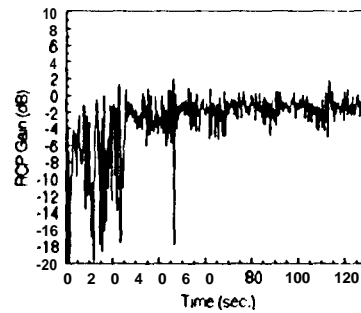
The work described here was motivated by the so-called “Virtual Spacecraft Concept” in which a **future spacecraft** is simulated initially by computer models of its subsystems. As the subsystems are developed in hardware, breadboard subsystems replace the computer models until ultimately, the simulation is nearly **all** hardware. In replacing a computer model of an antenna with hardware, we envision the antenna on the measurement range connected to the communications subsystem simulation. In simulating communication during spacecraft maneuvers, the positioner must move the antenna to make the transmit antenna execute the same trajectory in the **field** of view of the antenna as would the earth during the actual mission. This can require complex **multiaxis** motion. Moreover, to program this motion with high accuracy, one must know the positioner geometry with corresponding accuracy. Then in addition to compensating for positioner errors one may also compensate for changes in the distance to the transmit site during the motion so as to obtain true phase measurements representative of the actual earth spacecraft link.



## Link Simulation



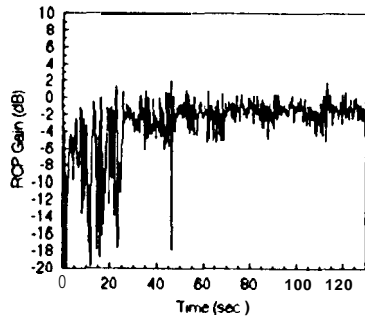
trajectory the rth the view the



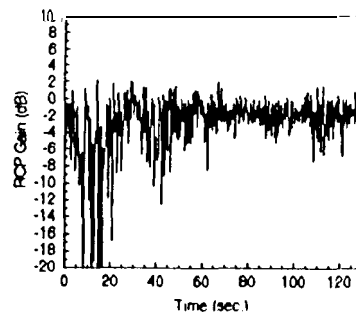
Calculated ● tenna gain during descent.

This is an example application which required a very complicated trajectory. This represents the **trajectory** of the 'earth in the field of view of the Mars Pathfinder lander low gain antenna during the descent to the surface of Mars. The graph on the right shows the computed right circular polarization signal magnitude variation due to motion along this trajectory.

## JPL Link Simulation (Continued)



*Calculated antenna gain during descent.*



*Measured antenna gain during descent.*

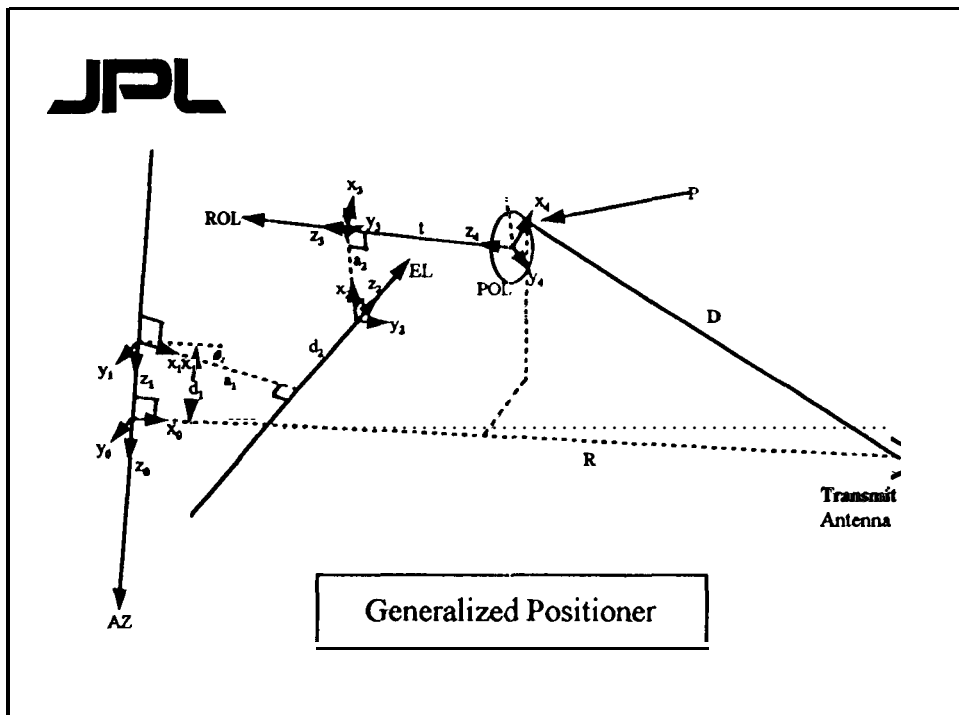
This shows a comparison between **the** computed gain variation and that measured using a crude mock-up **of the** lander. Qualitative agreement to the level expected considering the crudeness of the mock-up is evident.

# **JPL** Positioner Geometry

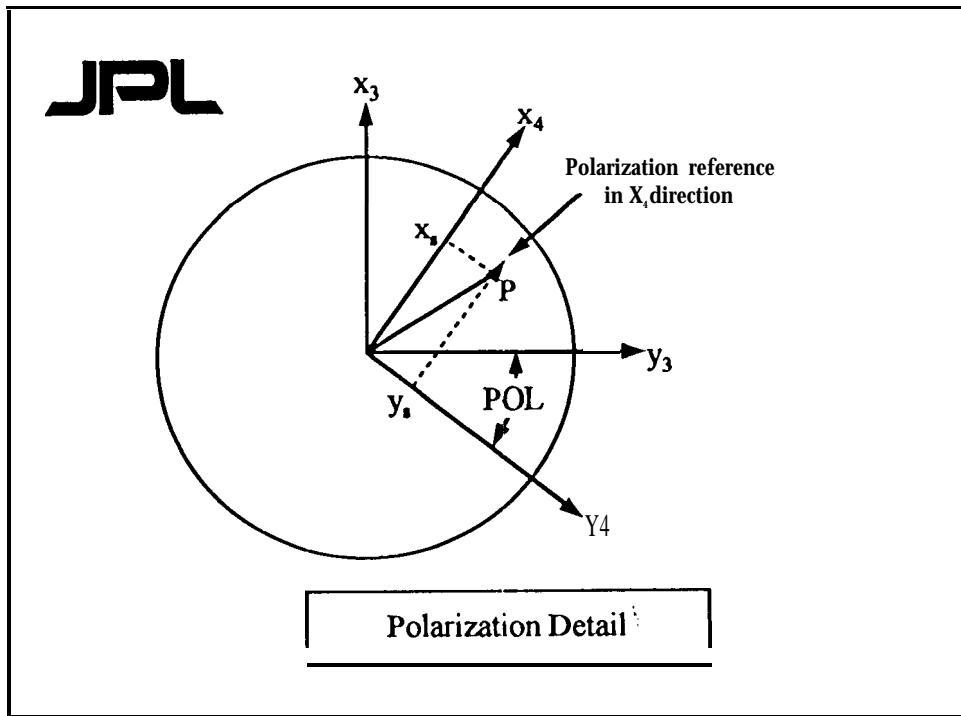
- A positioner is a robot.
  - Well established theoretical description available.
  - Standard system identification techniques applicable,
- Robotics provides:
  - Full description of **geometry**.
  - Means of programming arbitrary trajectories.
  - Means for compensating **for** defects.

The starting point of the development is that we recognize the positioner to be a robot and we apply the established theoretical description. This will provide a convenient and **compact** description of the positioner geometry and a convenient formalism in terms of **which** to describe trajectory programming **and** error compensation,





A robot is viewed as a series of links connected with hinges. The usual positioner actuators rotate these hinges. Associated with each hinge is a coordinate system with  $z$  axis along the hinge and  $x$  axis being the mutual perpendicular between that hinge axis and the next. The geometry is thus completely described by giving the perpendicular distances ( $a$ ) between the hinge axes, angles ( $\alpha$ ) between the hinge axes, and the distances along each axis between the intersections with the perpendiculars to the preceding and succeeding hinges ( $d$ ). In operation, the hinge angles ( $\theta$ ) are varied to produce the desired motion of the point  $P$ . The point  $P$  is described by a four vector in coordinate system 4 and a cascade of matrices describing transformation through coordinate systems 3, 2, and 1 lead to an overall transformation of the point  $P$  into the base coordinate system  $O$ .



Although our position **has** only three **axes**, we have added a fourth coordinate **system to describe the polarization of the** test antenna. Polarization is assumed to be linear and directed along the  $x$  axis of the fourth coordinate system. By properly setting the  $x$  and  $y$  coordinate of the antenna and the rotation angle,  $POL$ , one can achieve any linear polarization and any position desired.



## Robotics Model\*

$${}^0P = {}^0_4T \quad {}^4P = {}^0_1T \quad {}^1_2T \quad {}^2_3T \quad {}^3_4T \quad {}^4P$$

$${}^{i-1}_iT = \begin{bmatrix} \cos \theta_i & -\sin \theta_i & 0 & a_{i-1} \\ \sin \theta_i \cos \alpha_{i-1} & \cos \theta_i \cos \alpha_{i-1} & -\sin \alpha_{i-1} & -d_i \sin \alpha_{i-1} \\ \sin \theta_i \sin \alpha_{i-1} & \cos \theta_i \sin \alpha_{i-1} & \cos \alpha_{i-1} & d_i \cos \alpha_{i-1} \\ 0 & 0 & 0 & 1 \end{bmatrix}$$

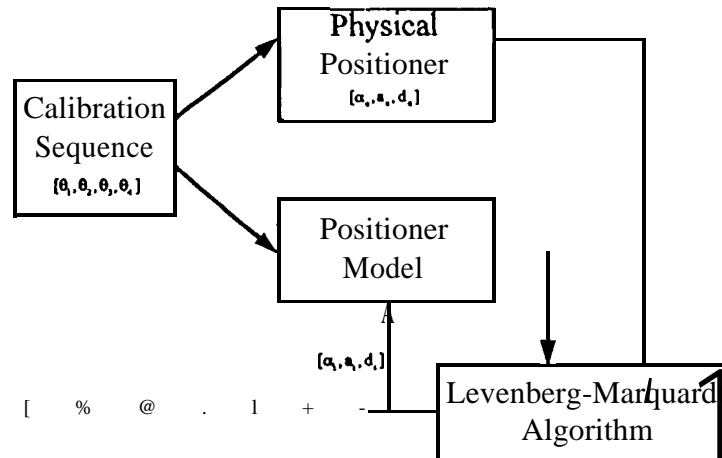
$${}^4P = \begin{bmatrix} x_e \\ y_e \\ 0 \\ 1 \end{bmatrix}$$

$$D = \sqrt{(R - x_0)^2 + y_0^2 + z_0^2}$$

\*J. J. Craig, Introduction to Robotics, Mechanics, and Control, Addison-Wesley, 1986.

This is the matrix description of the coordinate transformations. Each transformation matrix depends on the parameters of the given hinge and the rotation angle of that hinge. The product of all the transformation matrices gives a transformation matrix between coordinate systems 4 and 0.

## JPL System Identification

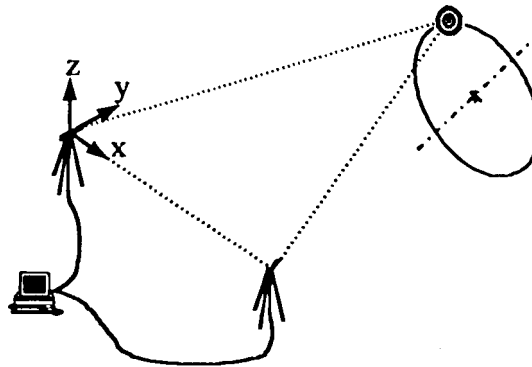


In terms of the preceding mathematical description one can develop a computer model of the positioner **with** variable geometrical parameters. It then becomes necessary to determine these parameters so as to mimic the actual positioner. This is the classical system identification problem. First, we **devise** a sequence of measurements (pattern cuts if you will) over which we measure **the** phase variation of the measured signal from, in this case, a circularly polarized transmit antenna. This yields a set of measured data. An important aspect of this is that the transmit antenna must be relatively close to the positioner for the measurement to be effective in **identifying** the parameters. In our case this measurement was done at 8.45 GHz and the distance was nominally 40 feet.

Having the above described data set in hand, we adjust the parameters of the computer model of the positioner until it produces a best fit to the data. This **was** accomplished using the standard **Levenberg-Marquardt** algorithm (Numerical Recipes).



## Optical Check



Some aspects of the resulting geometry were checked by using an optical measurement system (**Leica's** MANCAT System) consisting of two **theodolites** and a PC. The system establishes a coordinate system as shown and by sighting on a target from the two positions the computer algorithm can be used to provide the Cartesian coordinates of the target. By taking several such measurements and rotating one axis of the position in steps between measurements one obtains a set of points which are then fitted to a circle yielding the radius of the rotation circle, the coordinates of the center, and any desired number of points along the rotation axis. When this is done for all three axes of the positioner, the resulting data can be used to determine the angles and distances between the axes for comparison with the results derived from the phase data.

# JPL System Identification output

```

NUMBER OF DATA POINTS:      1086
ONE SIGMA PHASE ERROR:      20.0000    DEGREES
WAVELENGTH:                  1.40000    INCHES
CHISQUARE:                   1302.31
NUMBER OF ITERATIONS:        16

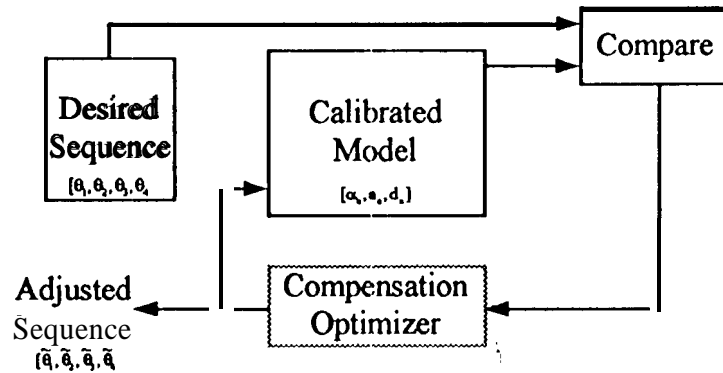
PARAMETER          INITIAL  EXPECTED  ESTIMATE  SIGMA  UNITS  FIXED
AZIMUTH OFFSET ANGLE:  0.00000  0.00000  -0.78086  0.02714  DEGREES
ELEVATION OFFSET ANGLE:  0.00000  0.00000  0.02979  0.01959  DEGREES
ROLL OFFSET ANGLE:      0.00000  0.00000  -0.31178  0.14100
POL OFFSET ANGLE:       0.00000  0.00000  0.00000  0.00000  DEGREES      x
AZ AXIS HEIGHT:         230.00000  251.25000  250.09961  0.02074  INCHES
SHIFT ALONG ELEVATION AXIS:  0.00000  0.66000  0.60284  0.01138  INCHES
AZ AXIS / EL AXIS ANGLE:  90.00000  89.72080  89.80862  0.01204  DEGREES
AZ AXIS / EL AXIS DISTANCE: 10.00000  21.71200  21.46828  0.01630  INCHES
EL AXIS / ROLL AXIS ANGLE: 0.00000  89.95320  89.49617  0.02195  DEGREES
EL AXIS / ROLL AXIS DISTANCE: 0.00000  0.04312  -0.30393  0.02261  INCHES
ROLL AXIS EXTENSION:     12.00000  8.00000  8.07395  0.01177  INCHES
X ANTENNA POSITION:      -20.00000  -30.75000  -10.01159  0.01167  INCHES
Y ANTENNA POSITION:       0.00000  0.00000  0.37934  0.07481  INCHES
TRANSMITTER DISTANCE:    461.29999  462.25000  463.29999  0.00000  INCHES      x
OVERALL PHASE OFFSET:    -346.79999  -346.79999  -346.79999  0.00000  WAVELENGTHS  x

```

This is an example of the output of the Levenberg-Marquardt based system identification algorithm. The **expected** parameters **marked** are those obtained from the optical **measurements** and the others by direct measurement or estimate. X's indicate those parameters held fixed during the optimization. There more subtleties involved in this process than can be discussed here. Suffice it to say that depending on the calibration measurements selected, various pairs of parameters can be dependent in such a manner that to get accurate results one or both of them must be fixed. Otherwise the algorithm may change both of them in a compensating manner and give erroneous results. Fortunately, such situations can be detected via the **covariance** matrix and thus can be avoided.



## Compensation Process



Once the positioner parameters have been “identified”, the robotics formalism can again be used to **adjust the programming of the positioner axis controllers to compensate for non-orthogonality of the axes and mis-calibration of the angles while determining the necessary axis rotations to achieve the desired measurement trajectory.** Moreover, during the measurement on that trajectory, the distance between the transmit and test antennas varies introducing artifacts in the phase measurements. This distance variation is now, however, known and can thus be removed leaving only the actual phase variation of the antenna under test.



## Concluding Remarks

- Positioners can be calibrated using robotics formalism.
- Robotics provides geometrical description.
- System identification techniques provide calibration,
- Robotics provides:
  - Trajectory programming.
  - Phase correction.
  - Error compensation.

We have shown how robotics formalism can be used to **provide** a convenient **description of range positioner geometry**. We have used this formalism in performing and processing calibration measurements to determine the geometrical parameters of the positioner and described how these parameters can be used in programming desired measurement trajectories, in phase correction, and in error compensation. As these techniques are refined, routine **re-calibration** of positioners can become an integral and automated part of antenna measurements. We believe that this is essential if accurate and **meaningful** phase data is to be taken particularly as the measurement frequency increases. Fortunately, the accuracy of this calibration technique increases with frequency. One might even consider doing the calibration at a higher frequency than that of the intended subsequent measurement to improve the accuracy.